

Ultra-efficient load introduction into composite structures

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BACKGROUND



- Lightweight metal-composite joints are essential for future aircraft
- State-of-the-art hybrid joining methods, such as fastening and riveting, are suboptimal:
 - Labour intensive
 - Weight penalty

OBJECTIVE(S)

- Develop efficient joining methods for metalcomposite structures that are:
 - Reliable (airworthy)
 - Damage tolerant (inspectable)
 - Dis-joinable (repairable)

WHY?

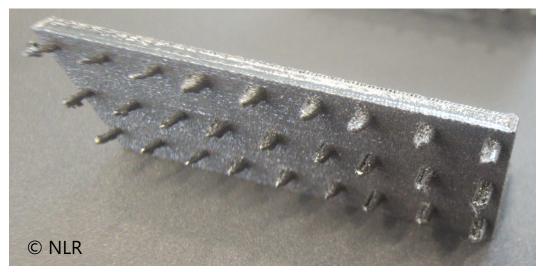


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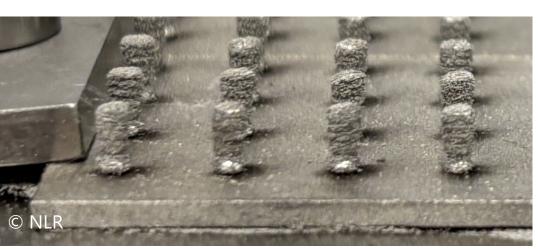


HOW?

- Study bonding of titanium (Ti6Al4V) and thermoplastic (LMPAEK) hybrid joint
- Study various joining methods:
 - Hot pressing (reference case)
 - Resistance welding
 - Induction welding
- Apply freeform additive manufacturing to tailor faying metal joint surface
 - Laser Powder Bed Fusion (LPBF) pins
 - Directed Energy Deposition (DED) pins



Pins produced by LPBF



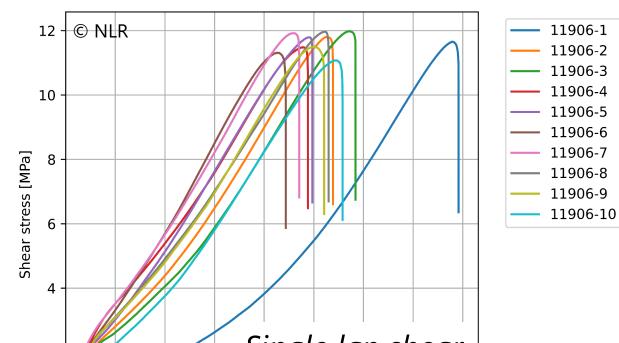
Pins produced by DED

- Increase sustainability
 - Better performance for lower weight
 - Double flush, no extending parts
 - Recyclable, no additional materials
- Increase competitiveness
 - Reduce labour and cost
 - Two-stage failure mode
 - Re-useable

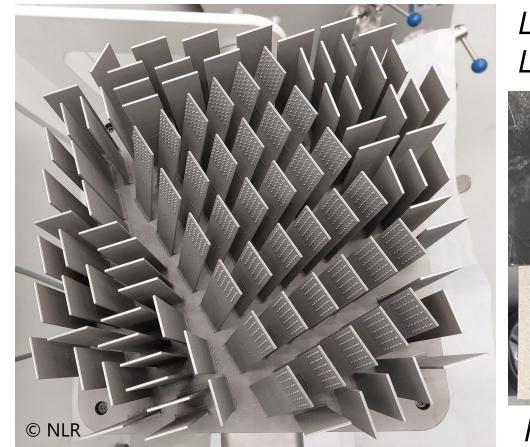
RESULTS



- Initial shear test results of hot-pressed hybrid titanium-LMPAEK joint
 - Without pins very weak shear strength (<1MPa)
 - With pins consistent failure around 11-12 MPa







LPBF build plate with 105 titanium Single Lap Shear (SLS) test specimens (left)



Single lap shear test with pins 0.000 0.002 0.004 0.006 0.008 0.010 0.012 0.014 0.016

• Future research activities

Strain [-]

Failed specimens with pins being both sheared off and pulled out of the LMPAEK

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- Collect more statistical data of bond strength
- Apply laser nano-structuring of bond surface
- Apply chemical treatment of bond surface
- Optimize interface geometry
- Work towards a demonstrator



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Acknowledgement

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[1] MBF 2300 series Installation and inspection specification: https://trsaero.com/ monogramaerospace/wp-content/uploads/sites/3/2018/03/MBF2301_CL_3_Installation.pdf